

# Data based modelling of electron fluxes at GEO and statistical wave models

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## **PROGRESS**

## Data based modelling of radiation belt electron fluxes at GEO



### **Radiation Belts**

Between 1985-2012 there have been 19 serious incidences

Five of which resulted in a total loss of the satellite

Irrecoverable Loss of Satellite

Telstar 401







Manufacture costs/satellite \$250 - \$350 M

Lost revenue /satellite ~\$ 150 M/year

Satellite lifetime 15-20 years

Galaxy 15



Unresponsive to ground control commands

Interfered with other communication satellites

Recovered after 1 year

### **Spacecraft in the Radiation Belts**

Courtesy- Orbital Sciences



### Galaxy 15

Geostationary communication satellite

Became unresponsive to commands after a small space weather event and began to drift

Galaxy 15's telecommunications remained fully functional

This could have interfered with the AMC-11 satellite that distributes television throughout the USA

## The effects of space weather: Radiation Belts

We need to be able to forecast the times when the radiation belt environment will be hazardous to the spacecraft to help satellite operators mitigate any issues arise with the spacecraft.

To forecast these events we need a reliable model of the radiation belts

#### Aims

Work Package 6 of PROGRESS is devoted to the development of models that are able to forecast the electron radiation in the radiation belts.

## ModellingFirst principlesvs.



### System identification approach

## ModellingFirst principlesvs.



### System identification approach

















### **System Identification**



## **System Identification**



#### Mapping the input to the output

- Neural Networks
- Genetic Algorithms
- Linear Prediction Filters
- NARMAX Physically Interpretable

Nonlinear

$$y(t) = F[y(t-1),...y(t-n_y),$$
  

$$u_1(t-1),...,u_1(t-n_{u_1}),...,$$
  

$$u_m(t-1),...,u_m(t-n_{u_m}),$$
  

$$e(t-1),...,e(t-n_e)] + e(t)$$

Nonlinear AutoRegressive

$$y(t) = F[y(t-1),...y(t-n_y),$$
  

$$u_1(t-1),...,u_1(t-n_{u_1}),...,$$
  

$$u_m(t-1),...,u_m(t-n_{u_m}),$$
  

$$e(t-1),...,e(t-n_e)] + e(t)$$

Nonlinear AutoRegressive Moving Average

$$y(t) = F[y(t-1),...y(t-n_y),$$
  

$$u_1(t-1),...,u_1(t-n_{u_1}),...,$$
  

$$u_m(t-1),...,u_m(t-n_{u_m}),$$
  

$$e(t-1),...,e(t-n_e)] + e(t)$$

Nonlinear AutoRegressive Moving Average with eXogenous inputs

$$y(t) = F[y(t-1), ..., y(t-n_y),$$

$$u_1(t-1), ..., u_1(t-n_{u_1}), ...,$$

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$$e(t-1),...,e(t-n_e)] + e(t)$$

NARMAX Model:

- Nonlinear Function *F*. e.g. Polynomial, Wavelets, etc.
  - Degree of polynomial
  - Type of wavelet
- Inputs
- System lags

#### Nonlinear AutoRegressive Moving Average with eXogenous inputs

$$y(t) = F[y(t-1),...y(t-n_y),$$

$$u_1(t-1),...,u_1(t-n_{u_1}),...,$$

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$$e(t-1),...,e(t-n_e)] + e(t)$$

#### NARMAX Model:

- Nonlinear Function *F*. e.g. Polynomial, Wavelets, etc.
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Polynomial

- FROLS algorithm Involves three stages
  - 1. Structure selection: Error Reduction Ratio (ERR)
  - 2. Coefficient estimation
  - 3. Model validation















#### NARMAX FROLS

#### **Coefficient Estimation**



#### NARMAX FROLS

#### **Coefficient Estimation**



Squares

#### NARMAX FROLS

#### **Coefficient Estimation**



#### **Electron Flux Models**

A separate NARMAX model was developed for the >800 keV and >2 MeV energies using:

**Output Data** GOES Electron Fluxes *J* Lags: 24 hours, 48 hours

#### **Input Data**

Solar wind Velocity *V*, Density *n*, the Dst Index *Dst*, z IMF *Bz*, and the time IMF was southward per day  $\tau_{Bz}$ . Lags: 24 hours, 48 hours

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F is a third degree polynomial

$$\begin{aligned} f(t) &= F[J(t-24h), J(t-48h), \\ V(t-24h), V(t-48h), \\ n(t-24h), n(t-48h), \\ B_z(t-24h), B_z(t-48h), \\ \tau_{B_z}(t-24h), \tau_{B_z}(t-48h), \\ Dst(t-24h), Dst(t-48h), \\ e(t-24h), e(t-48h)] \end{aligned}$$

#### **Electron Flux Models - Performance**

The performance of the model was assessed using the Correlation Coefficient (CC)

$$CC = \frac{\sum_{t=1}^{N} \left[ \left( y(t) - \overline{y}(t) \right) \left( \hat{y}(t) - \overline{\hat{y}}(t) \right) \right]}{\sqrt{\sum_{t=1}^{N} \left[ \left( y(t) - \overline{y}(t) \right)^{2} \right] \sum_{t=1}^{N} \left[ \left( \hat{y}(t) - \overline{\hat{y}}(t) \right)^{2} \right]}}$$

and Prediction Efficiency (PE)

$$PE = 1 - \frac{\sum_{t=1}^{N} \left[ \left( y(t) - \hat{y}(t) \right)^{2} \right]}{\sum_{t=1}^{N} \left[ \left( y(t) - \overline{y}(t) \right)^{2} \right]}$$

Where y(t) is the measured output at time t,  $\hat{y}$  is the forecast output, N is the length of the data and the bar indicates the mean.

### **Electron Flux Model – SNB<sup>3</sup>GEO**

>800 keV Electron flux model at geosynchronous orbit



>2 MeV Electron flux model at geosynchronous orbit

PE = 0.786 and CC = 0.894 for over 26 months of data between 14/04/2010 to 30/06/2012



#### **Electron Flux Model – SNB<sup>3</sup>GEO**



#### **Electron flux – SNB<sup>3</sup>GEO**

#### NOAA-REFM vs. SNB<sup>3</sup>GEO



## Electron flux – SNB<sup>3</sup>GEO

#### NOAA-REFM vs. SNB<sup>3</sup>GEO

Balikhin et al. [2016], Space Weather

#### Fluxes

Model	Correlation	PE
REFM	0.73	-1.31
SNB <sup>3</sup> GEO	0.82	0.63

#### log<sub>10</sub>(Fluxes)

Model	Correlation	PE
REFM	0.85	0.70
SNB <sup>3</sup> GEO	0.89	0.77

March 2<sup>nd</sup>, 2012 - January 1<sup>st</sup> 2014.




# **Electron Flux Models - Performance**

#### Heidke Skill score

Event Forecast	Event Observed		
	Yes	No	Marginal Total
Yes	а	b	a + b
No	с	d	c + d
Marginal Total	a + c	b + d	a+b+c+d=n

$$HSS = \frac{2(ad - bc)}{[(a+c)(c+d) + (a+b)(b+d)]}$$

## **Electron Flux Models - Performance**

#### **NOAA-REFM**

Fluence (cm <sup>-2</sup> sr <sup>-1</sup> d <sup>-1</sup> )	>`	10 <sup>8</sup>	>1	0 <sup>8.5</sup>	>	>10 <sup>9</sup>
REFM HSS	0.	666	0.	482	(	).437
Observation	Yes	No	Yes	No	Yes	No
Forecast						
Yes	<i>x</i> = 86	<i>z</i> = 22	<i>x</i> = 23	<i>z</i> = 22	<i>x</i> = 4	<i>z</i> = 7
No	<i>y</i> = 43	<i>w</i> = 510	<i>y</i> = 21	w = 595	<i>y</i> = 3	<i>w</i> = 647

#### SNB<sup>3</sup>GEO

Fluence (cm <sup><math>-2</math></sup> sr <sup><math>-1</math></sup> d <sup><math>-1</math></sup> )	>1	10 <sup>8</sup>	>1	0 <sup>8.5</sup>	2	>10 <sup>9</sup>
SNB <sup>3</sup> GEO HSS	0.738		0.634		0.612	
Observation	Yes	No	Yes	No	Yes	No
Forecast						
Yes	<i>x</i> = 106	<i>z</i> = 33	<i>x</i> = 31	<i>z</i> = 19	<i>x</i> = 4	<i>z</i> = 2
No	<i>y</i> = 23	<i>w</i> = 499	<i>y</i> = 13	<i>w</i> = 598	<i>y</i> = 3	<i>w</i> = 652

# **Electron Flux Models: Low energies**

A separate NARMAX model was developed for each of the 5 low energies (30-50 keV, 50-100 keV, 100-200 keV, 200-350 keV, 350-600 keV) using:

#### **Output Data**

GOES Electron Fluxes Lags: 24 hours, 48 hours

#### **Input Data**

Solar wind Velocity *V*, Density *n*, Pressure *p*, the Dst Index *Dst*, and southward IMF *B* Lags: 2 hours, 3 hours,..., 48 hours

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#### **Output Data**

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Solar wind Velocity *V*, Density *n*, Pressure *p*, the Dst Index *Dst*, and southward IMF *B* Lags: 2 hours, 3 hours,..., 48 hours

F is a fourth degree polynomial

$$\begin{split} J(t) &= F[J(t-24h), J(t-48h), \\ v(t-2h), v(t-3h), ..., v(t-48h), \\ n(t-2h), n(t-3h), ..., n(t-48h), \\ p(t-2h), p(t-3h), ..., p(t-48h), \\ Dst(t-2h), Dst(t-3h), ..., Dst(t-48h), \\ B(t-2h), B(t-3h), ..., B(t-48h), \\ e(t-24h), e(t-48h)] + e(t) \end{split}$$

Model	Forecast Horizon (hours)	PE (%)	CC (%)	Period
40-50 keV	10	66.9	82.0	01.03.2013- 28.02.2015
50-100 keV	12	69.2	83.5	01.03.2013- 28.02.2015
100-200 keV	16	73.2	85.6	01.03.2013- 28.02.2015
200-350 keV	24	71.6	84.9	01.03.2013- 28.02.2015
350-300 keV	24	73.6	85.9	01.03.2013- 28.02.2015
> 800 keV	24	72.1	85.1	01.01.2011- 28.02.2015
>2MeV	24	82.3	90.9	01.0.12011- 28.02.2015

# **Forecast Horizon of NARMAX models**

The amount of time that the NARMAX model is able to forecast into the future is dependent on the minimum exogenous lag within the final NARMAX model.

For example, if the minimum exogenous lag within the NARMAX model is a velocity value 10 hours ago

$$J(t) = aV(t-10) + \dots$$

Where a is the coefficient, then if we know the velocity at the present time t, then we can calculate an estimate of the electron flux, J, at time t+10 hours (a 10 hour ahead forecast)

$$J(t+10) = aV(t) + \dots$$

#### **Model Performance Figures**



### **Model Performance Figures**



### **Model Performance Figures**



# **Real-time operation**





## **Physics Based Models**



The evolution of the radiation belt electrons can be modelled by the bounce-averaged Fokker-Planck equation [Schulz and Lanzerotti, 1974]:

$$\begin{split} \frac{\partial f}{\partial t} &= L^{*2} \frac{\partial}{\partial L^*} \left| \frac{1}{\mu, J} \frac{1}{L^{*2}} D_{L^*L^*} \frac{\partial f}{\partial L^*} \right|_{\mu, J} + \frac{1}{p^2} \frac{\partial}{\partial p} \right|_{\alpha_0, \mathcal{L}} \\ & \cdot p^2 \left( D_{pp} \frac{\partial}{\partial p} \left|_{\alpha_0, \mathcal{L}} f + D_{p\alpha_0} \frac{\partial}{\partial \alpha_0} \right|_{p, \mathcal{L}} f \right) + \frac{1}{T(\alpha_0) \sin(2\alpha_0)} \frac{\partial}{\partial \alpha_0} \left|_{p, \mathcal{L}} f \right. \\ & \left. \cdot T(\alpha_0) \sin(2\alpha_0) \left( D_{\alpha_0 \alpha_0} \frac{\partial}{\partial \alpha_0} \left|_{p, \mathcal{L}} f + D_{\alpha_0 p} \frac{\partial}{\partial p} \right|_{\alpha_0, \mathcal{L}} f \right) + \frac{f}{\tau}, \end{split}$$

## **Physics Based Models**



The evolution of the radiation belt electrons can be modelled by the bounce-averaged Fokker-Planck equation [Schulz and Lanzerotti, 1974]:

$$\begin{split} \frac{\partial f}{\partial t} &= L^{*2} \frac{\partial}{\partial L^*} \left| \frac{1}{L^{*2}} D_{L^*L^*} \frac{\partial f}{\partial L^*} \right|_{\mu,J} + \frac{1}{p^2} \frac{\partial}{\partial p} \right|_{\alpha_0, \mathcal{L}} \\ & \cdot p^2 \left( D_{pp} \frac{\partial}{\partial p} \left|_{\alpha_0, \mathcal{L}} f + D_{p\alpha_0} \frac{\partial}{\partial \alpha_0} \right|_{p, \mathcal{L}} f \right) + \frac{1}{T(\alpha_0) \sin(2\alpha_0)} \frac{\partial}{\partial \alpha_0} \left|_{p, \mathcal{L}} f \right. \\ & \left. \cdot T(\alpha_0) \sin(2\alpha_0) \left( D_{\alpha_0 \alpha_0} \frac{\partial}{\partial \alpha_0} \left|_{p, \mathcal{L}} f + D_{\alpha_0 p} \frac{\partial}{\partial p} \right|_{\alpha_0, \mathcal{L}} f \right) + \frac{f}{\tau}, \end{split}$$

These models, such as Versatile Electron Radiation Belt (VERB) model employ numerical codes that involve finding solutions of the diffusion equations.

# **Diffusion Coefficients**

Many approaches have been developed to calculate the diffusion coefficients, all of which require models of various waves.

For example, the VERB code employs statistical wave models for Lower Band Chorus (LBC), Hiss and Equatorial MagnetoSonic (EMS) waves.

Currently, the statistical models of the waves distributions employ wave measurements on various spacecraft, which are parameterized by the location of observations and current values for geomagnetic indices neglecting solar wind measurements and geomagnetic evolution.

#### Aims

Work Package 4 of PROGRESS aims to determine the influential parameters (solar wind and geomagnetic indices) that control the wave amplitude distribution at particular locations and then redevelop the statistical wave models

# **Diffusion coefficients**

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#### Statistical wave models

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Aryan et al., JGR, 2014

Figure 2. Equatorial wave intensity of lower band chorus as a function of  $L^*$ , MLT and geomagnetic activity for each of the five satellites.

#### Meredith et al., JGR, 2012

## How to identify wave control parameters



What parameters influence the waves in the radiations belts?

How to identify these parameters?

A simple quadratic system

$$y(t) = x^2(t-1) + e(t)$$

Where the output *y* at time *t* is a function of zero mean signal *x* and noise *e* 

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A simple quadratic system

$$y(t) = x^2(t-1) + e(t)$$

Where the output *y* at time *t* is a function of zero mean signal *x* and noise *e* 

Calculate the correlation function:

$$\phi_{xy}(\tau) = \frac{\sum_{t=1}^{N} \left[ \left( y(t-\tau) - \bar{y} \right) \left( x(t) - \bar{x} \right) \right]}{\sqrt{\sum_{t=1}^{N} \left[ \left( y(t) - \bar{y} \right)^2 \right] \sum_{t=1}^{N} \left[ \left( x(t) - \bar{x} \right)^2 \right]}}$$

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$$y(t) = x^2(t-1) + e(t)$$

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# NARMAX FROLS ERR

# Better to use techniques that are able to account for nonlinear systems, such as NARMAX FROLS ERR



# NARMAX FROLS ERR

A simple quadratic system

$$y(t) = x^2(t-1) + e(t)$$

Where the output *y* at time *t* is a function of zero mean signal *x* and noise *e* 

NARX model:

$$y(t) = F[y(t-1), y(t-2), x(t-1), x(t-2), v(t-3), v(t-1), v(t-2), v(t-3)]$$

F as a third degree polynomial

Where *v* was a random variable

TERM	ERR (%)
$x^2(t-1)$	98.6

## Wave data





# ERR analysis of radiation belt waves

#### **Output Data**

Wave intensity,  $B_w$ , for each MLT, L bin From THEMIS, Cluster and Double Star

 Input Data
  $B_w(L, MLT, t) = F[V(t-0), V(t-2), ..., V(t-20),$  

 Solar wind Velocity V,
 n(t-0), n(t-2), ..., n(t-20), 

 Density n,
 p(t-0), p(t-2), ..., p(t-20), 

 Pressure p,
  $B_B(t-0), B_B(t-2), ..., B_B(t-20)]$  

 and IMF factor  $B_B$  Lags: 0 hours, 2 hours, 4 hours,..., 20 hours

N.B. *F* contains no autoregressive or moving average terms as wave data is too sparse







LBC Waves: SW inputs, v15 data: All Spacecraft















Hiss Waves: SW inputs, v15 data: All Spacecraft





















# ERR analysis of radiation belt waves

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Wave intensity,  $B_w$ , for each MLT, L bin From THEMIS, Cluster and Double Star

Input Data	$B_w(L, MLT, t) = F[V(t-0), V(t-2),, V(t-20),$
Solar wind Velocity V,	n(t-0), n(t-2),, n(t-20),
Density <i>n</i> ,	p(t-0), p(t-2),, p(t-20),
Pressure <i>n</i> .	$B_B(t-0), B_B(t-2),, B_B(t-20),$
IME factor $B_{-}$	Dst(t-0), Dst(t-2),, Dst(t-20),
Det index $D_B$ ,	AE(t-0), AE(t-2),, AE(t-20)]
$Dst \operatorname{Index} Dst,$	
And AE index AE.	

Lags: 2 hours, 4 hours,..., 20 hours

N.B. *F* contains no autoregressive or moving average terms as wave data is too sparse

#### Quadratic F

LBC Waves: SW-C-GI inputs, v16 data: All Spacecraft




## ERR analysis of radiation belt waves: Hiss waves



Hiss Waves: SW-C-GI inputs, v15 data: All Spacecraft





## ERR analysis of radiation belt waves: EMS waves

## Quadratic F

EMS Waves: SW-C-GI inputs, v15 data: All Spacecraft









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