



SPACESTORM



Understanding the radiation environment in the Earth's inner magnetosphere

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and SPACESTORM and PROGRESS teams

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4th Cluster and THEMIS workshop, 7-12 November 2016, Palm Springs, CA, USA



EU FP7 SPACESTORM project overview

<http://www.spacestorm.eu/>

SPACESTORM is a collaborative Project funded by the European Union's 7th Framework Programme to model space weather events and mitigating their effects on satellites.

The project builds on the forecasting of space weather started by the FP7 SPACECAST project.

The SPACESTORM consortium consists of five partners:

- (1) Natural Environment Research Council – British Antarctic Survey (NERC-BAS), UK
- (2) Ilmatieteen Laitos (FMI, Finnish Meteorological Institute), Finland
- (3) DH Consultancy BVBA (DHC), Belgium
- (4) University of Surrey – Surrey Space Centre (SSC), UK
- (5) Office National D'Etudes et de Recherches Aerospatiales (ONERA), France

Goal

To model severe space weather events and mitigate their effects on satellites by developing better mitigation guidelines, forecasting, and by experimental testing of new materials and methodologies to reduce vulnerability.



Modelling of 30 years of radiation belts

- BAS Radiation Belt Model: diffusion equation for the drift averaged phase-space density, includes (1) Radial transport, (2) Wave-particle interactions (Plasmaspheric hiss, Lightning generated whistlers, Upper band, lower band and low-frequency chorus, EMIC waves), (3) Loss to the atmosphere, (4) Loss to the magnetopause.
- >2 MeV electron flux from GOES provides outer boundary condition
Model requires whole energy spectrum at a fixed L^*
Data provides one integral energy at varying L^* (diurnal variation)
- Asynchronous regression [O'Brian et al., 2001] removes diurnal variation
Maps the flux measurement at any MLT, to the flux that would be measured by the same instrument at a fixed, reference local time
So the >2 MeV flux is mapped to a fixed L^*
- Use activity dependent spectra fitted at >2 MeV to get whole spectrum
Developed a set of spectra from 150keV, 275 keV, 475 keV, >800 keV and >2 MeV channels on GOES 15
Use these to get spectrum at fixed L^* from the mapped >2 MeV flux

30 years simulation

Long term variability

Most intense

in declining phase

1993-1994, 2003-2005

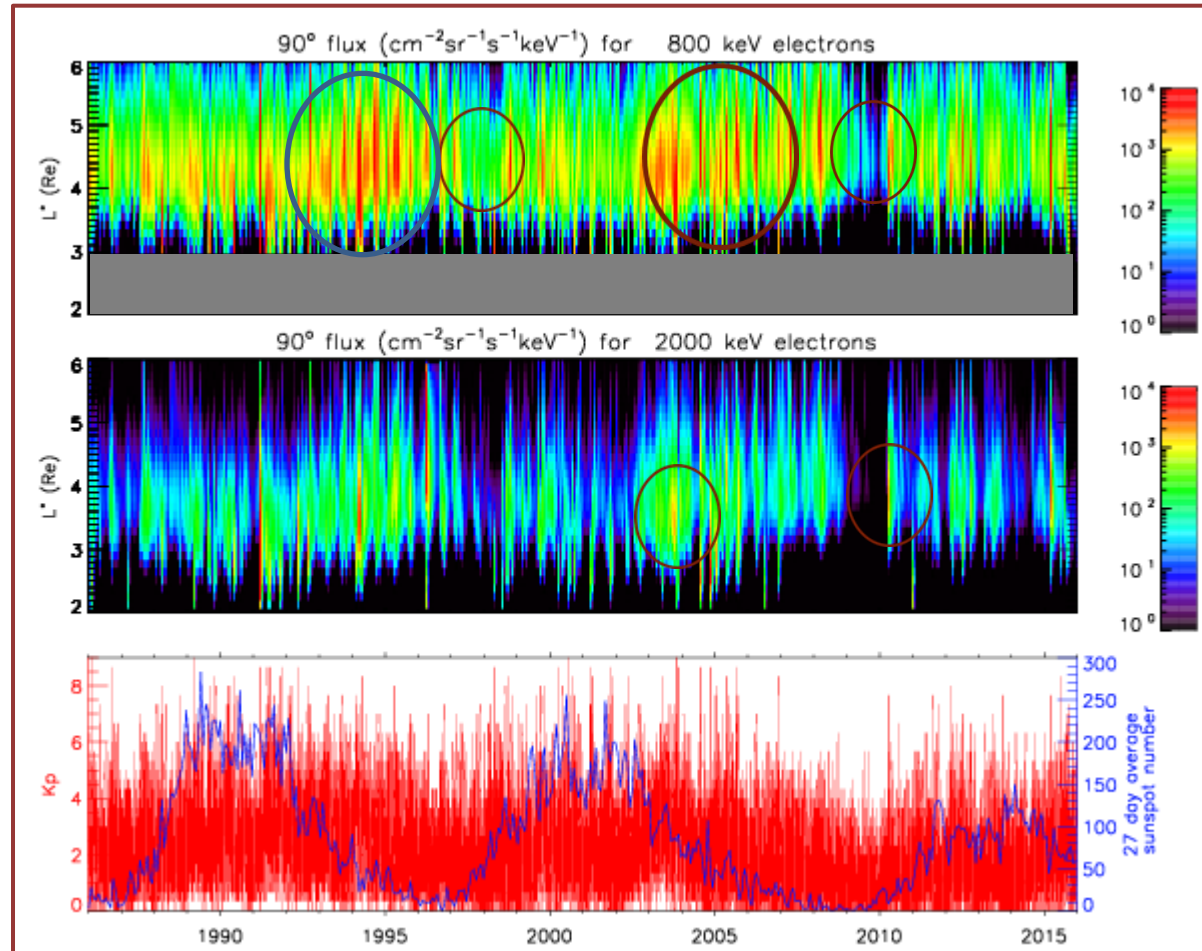
Quiet start to new cycle

1998, 2009

2 MeV at $L^*=3.5$

peak flux can be
several orders of
magnitude different

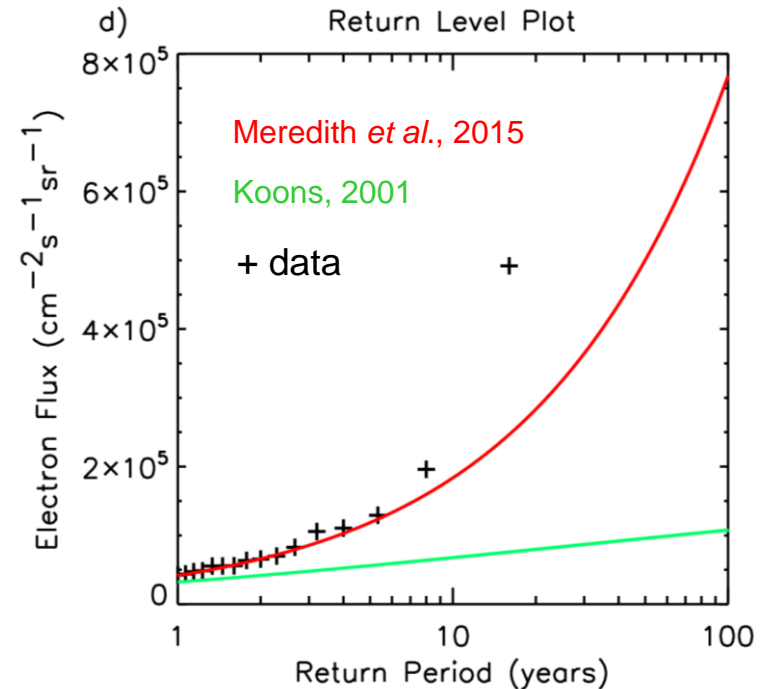
for extended periods



Extreme value analysis

- BAS conducted an extreme value analysis of 19.5 years of $E > 2$ MeV electron data from the GOES satellites at geosynchronous orbit. It was found

- The 1 in 10 year flux at GOES West was $1.84 \times 10^5 \text{ cm}^{-2} \text{ s}^{-1} \text{ sr}^{-1}$
- The 1 in 100 year flux at GOES West was $7.68 \times 10^5 \text{ cm}^{-2} \text{ s}^{-1} \text{ sr}^{-1}$
- The 1 in 10 year and 1 in 100 year fluxes at GOES West were factors of 2.7 and 7 times those estimated by [Koons \[2001\]](#)

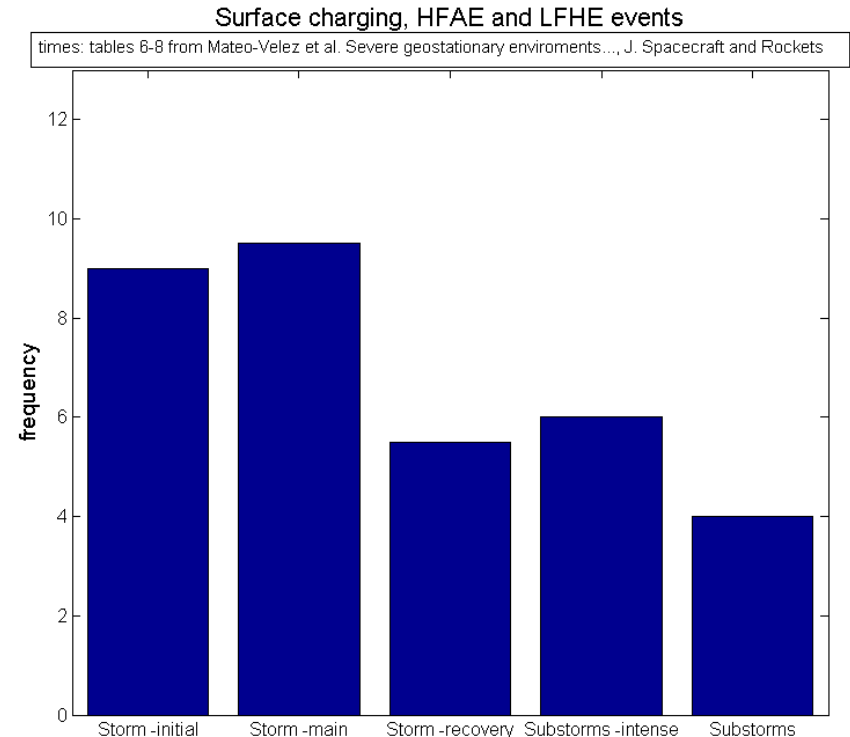


Importance of keV electrons in the inner magnetosphere

Surface charging events vs. geomagnetic conditions

- The distribution of low energy electrons population (10 to few hundreds of keV) constitutes the **seed population** further accelerated to MeV energies, critically important for **radiation belt** dynamics
- keV electrons are responsible for **surface charging** can cause significant damage and spacecraft anomalies
- **Louis Lanzerotti: Space weather is more than storms**
It is **NOT** necessary to have even a **moderate storm for significant surface charging** event to happen
- The electron flux at the keV energies varies significantly with geomagnetic activity
variations on time scales of minutes!

No averaging over an hour/day/orbit!



Matéo Véléz et al., Severe geostationary environments: from flight data to numerical estimation of spacecraft surface charging, *Journal of Spacecraft and Rockets*, 2016.

It is not easy to model (nowcast) and forecast low energy electrons

- Following low energy electrons in large-scale **magnetic and electric fields**:
Correct models for these fields are extremely hard to develop
 - Specification of a correct **initial conditions in the plasma sheet** is very nontrivial
 - **Coefficients for radial diffusion** when electrons move from the plasma sheet (10 Re) to inner regions (<6 Re) are far from being exact.
 - How to introduce low energy electrons' losses correctly? Electron lifetimes due to interactions with chorus and hiss, other waves, are they important?

- **MAIN FACTOR: SUBSTORMS.**

Substorms play a significant role in keV **electron transport and energy increase.**

How to include them properly?

- Like electromagnetic pulse? [*Li et al.*, 1998; *Zaharia et al.*, 2000; *Sarris et al.*, 2002; *Ganushkina et al.*, 2005, 2013; *Gabrielse et al.*, 2012, 2014] What are the parameters? Most probably, not the amplitude. Location? MLT-width?
- Do we need different representations for different types of substorms (isolated substorms, storm-time substorms)?
- Low energy electrons (at geostationary) are not organized by AE, KP-organization misses dynamics, IMF BZ and V_{sw} are main parameters.

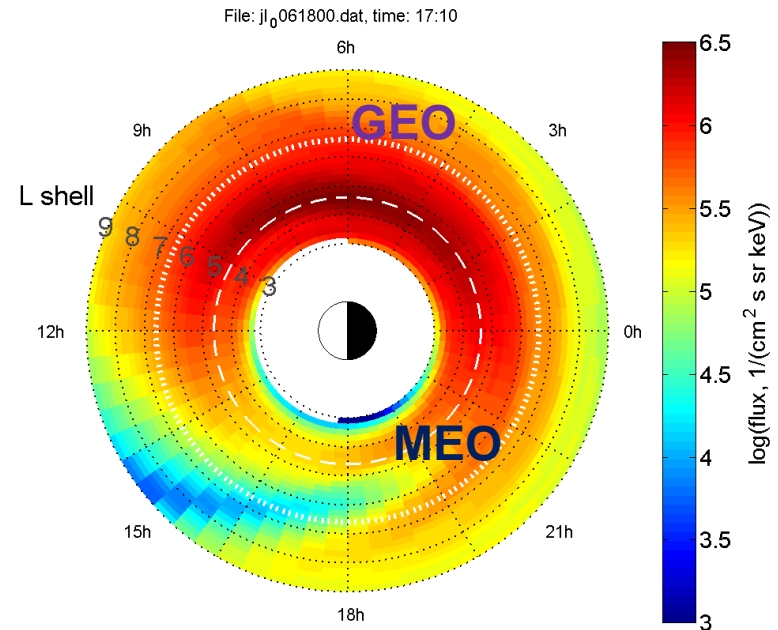
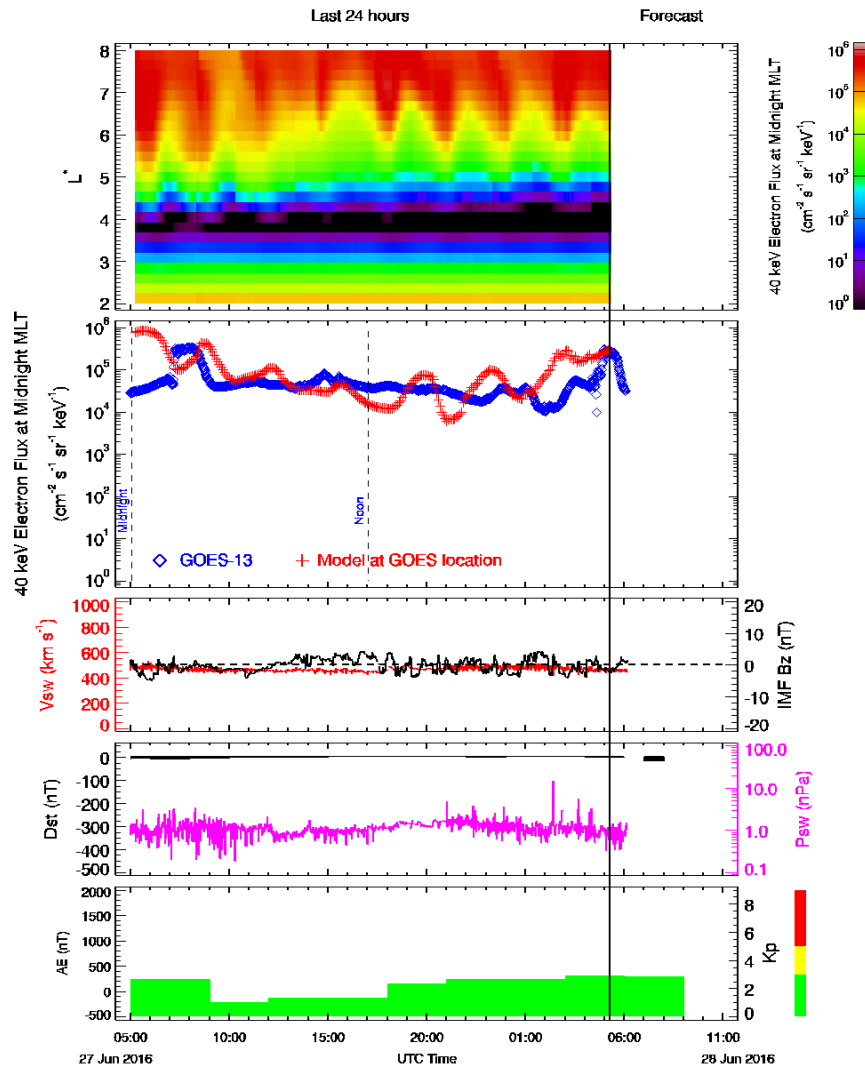
Present IMF and SW dependent models fail to represent the observed peaks associated with substorm activity

IMPTAM compared to GOES MAGED

40 keV e- fluxes

IMPTAM: traces electrons (< 200 keV) with arbitrary pitch angles (**drift approximation**) from the plasma sheet to the inner L-shells in time-dependent magnetic and electric fields

Taken into account: **radial diffusion and electron losses** as convection outflow and pitch angle diffusion by the **electron lifetimes**



<http://csem.engin.umich.edu/tools/imptam>
imptam.fmi.fi

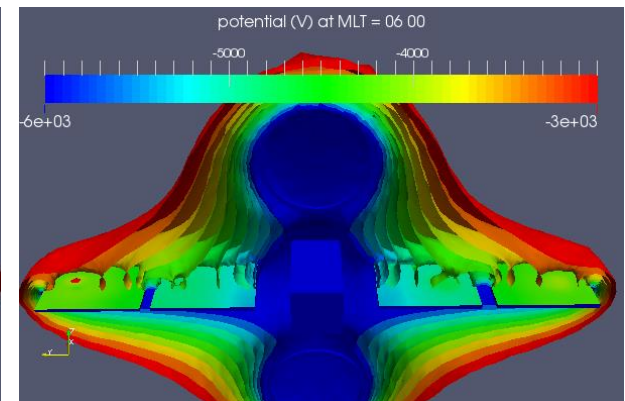
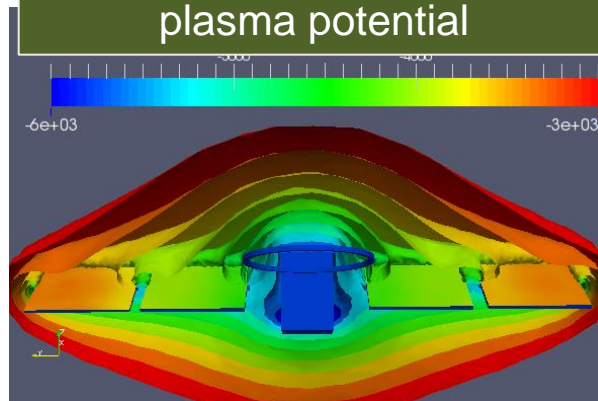
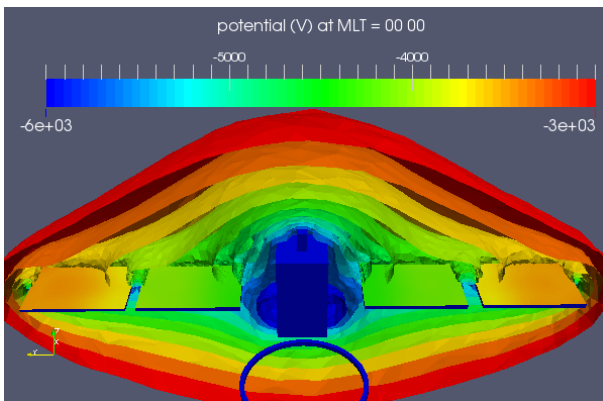
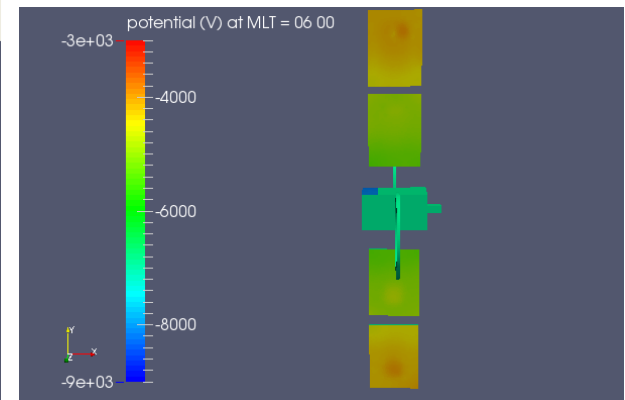
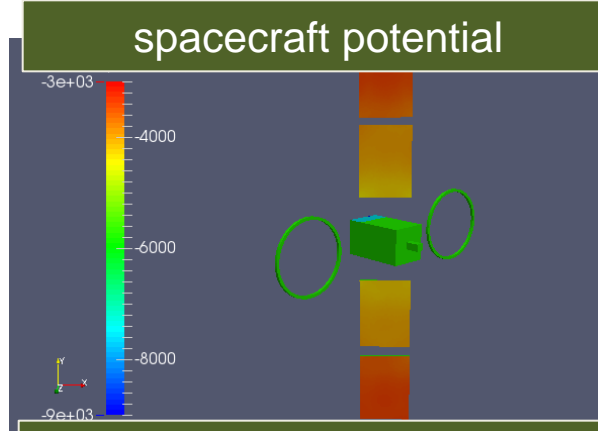
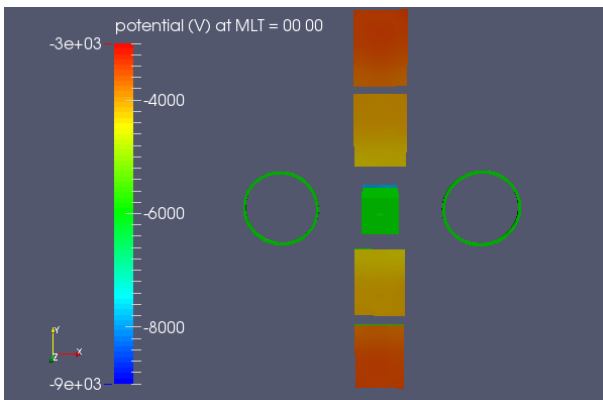
Ganushkina et al., JGR, 2013, 2014; Space Weather, 2015.

Surface charging risk assessment

00 MLT

03 MLT

06 MLT



Mateo-Velez et al 2016, 14th SCTC

With the same environment: Charging risk are more important at 06 MLT

The spacecraft attitude and of the area of conductive materials exposed to sunlight are very important

ONERA results

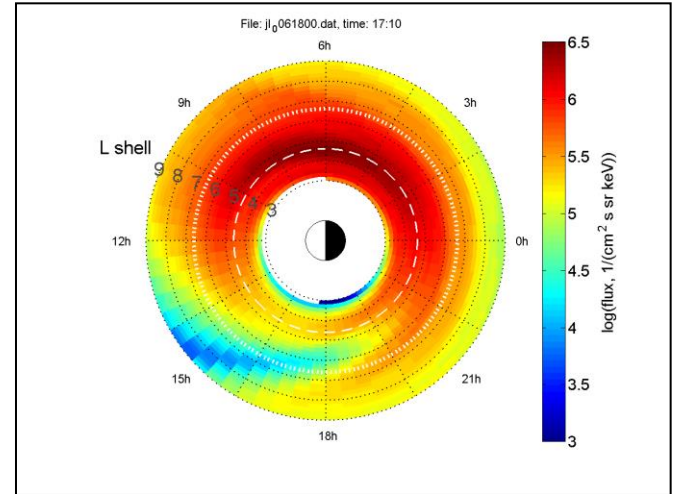


Charging at MEO

Very few data available

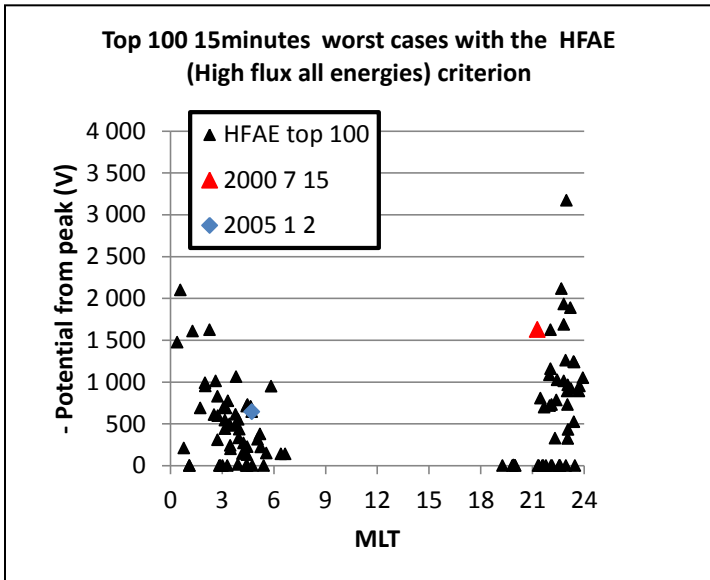
Method to obtain MEO worst case flux

1. Select dates of charging events at LANL (list provided by ONERA)
2. Use the IMPTAM (FMI) to transport electrons from GEO (LANL) to MEO L = 4.6
3. Select time and position of worst case electron fluxes at MEO

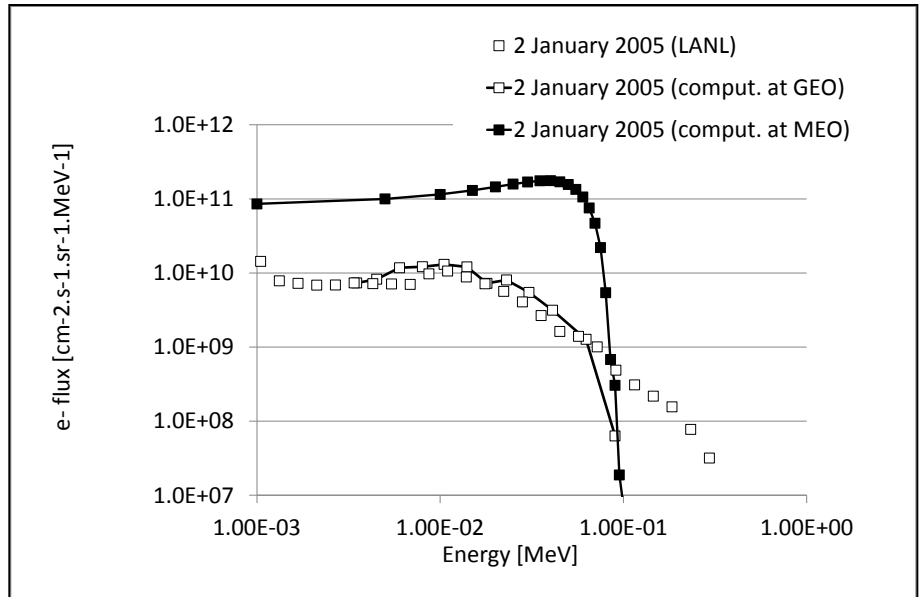


2. IMPTAM

Mateo-Velez et al 2016, 14th SCTC



1. GEO LANL (courtesy of CNES)



3. Specification at MEO

EU H2020 PROGRESS project overview

PRediction Of Geospace Radiation Environment and Solar wind parameterS

ssg.group.shef.ac.uk/progress/html

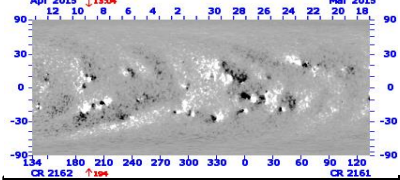
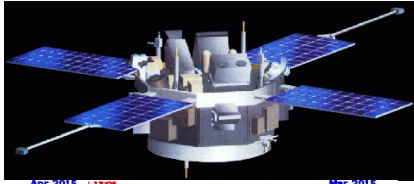
Participants:

- (1) School of Mathematics and Statistics, Department of Automatic Control and Systems Engineering, The University of Sheffield, Sheffield, UK
- (2) Finnish Meteorological Institute, Helsinki, Finland
- (3) University of Warwick, Coventry, UK
- (4) University of Michigan, Ann Arbor, MI, USA
- (5) Space Research Institute, Kiev, Ukraine
- (6) Laboratoire de Physique et Chimie de l'Environnement et de l'Espace, Orleans, France
- (7) Swedish Institute of Space Physics, Lund, Sweden
- (8) Helmholtz Centre Potsdam GFZ, Potsdam, Germany

The **overall aim** of the project PROGRESS is to exploit the available spacecraft and ground based data combined with state of art data assimilation methodologies in order to develop an accurate and reliable forecast of space weather hazards.



PROGRESS Overview



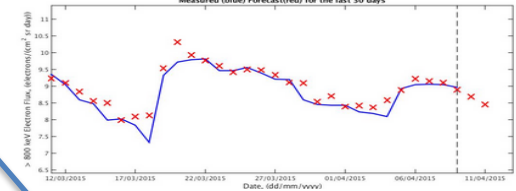
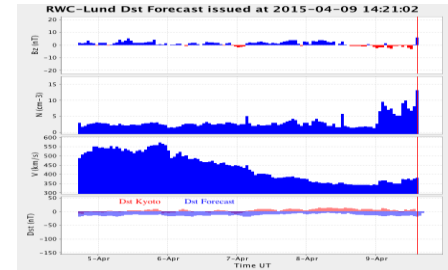
Solar wind propagation from Sun to L1 (AWSoM/SWIFT)

Development of new statistical models

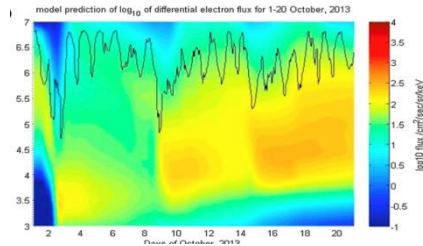
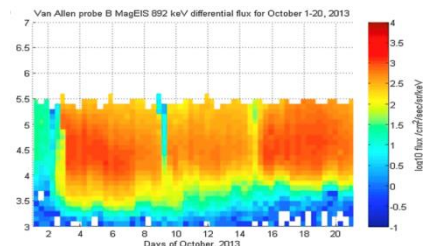
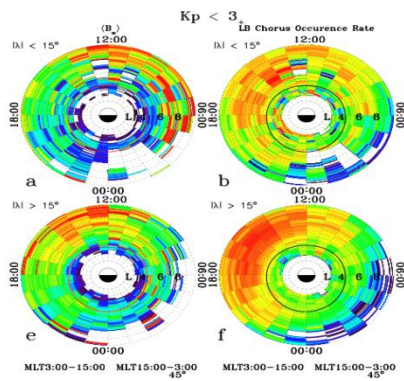
Low energy electron model

Forecast of the Evolution of Geomagnetic indices

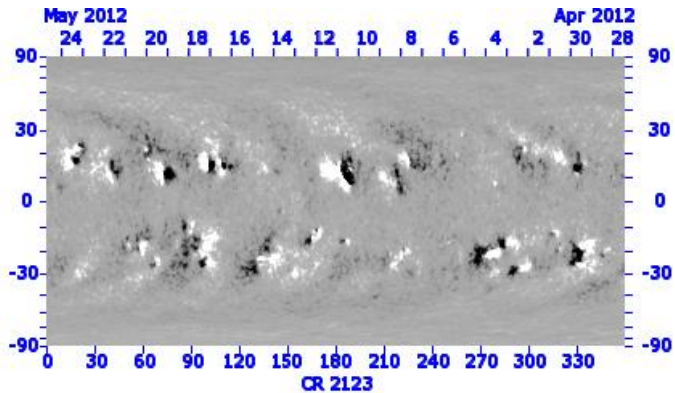
Forecast of the high energy electron environment



Fusion of forecast tools



Prediction of L1 data from GONG



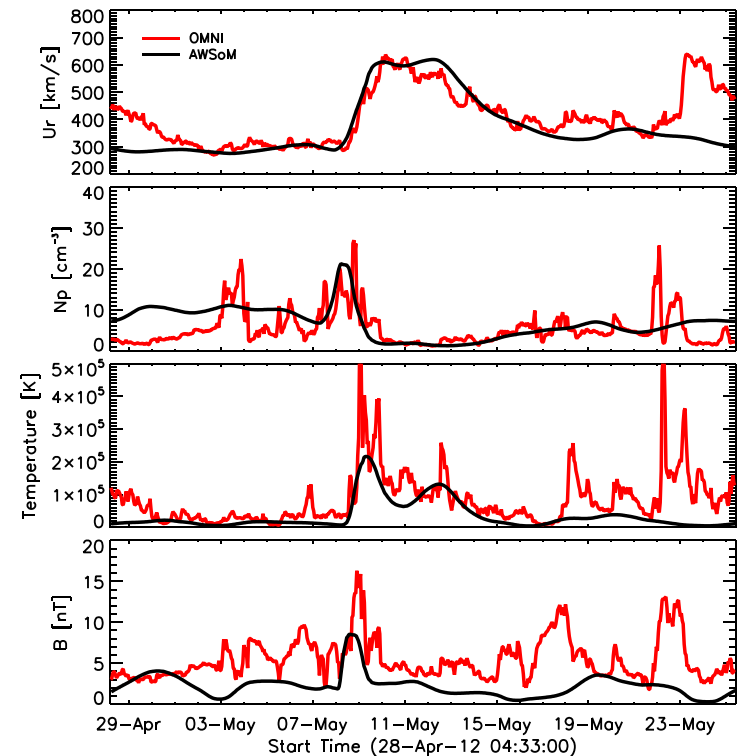
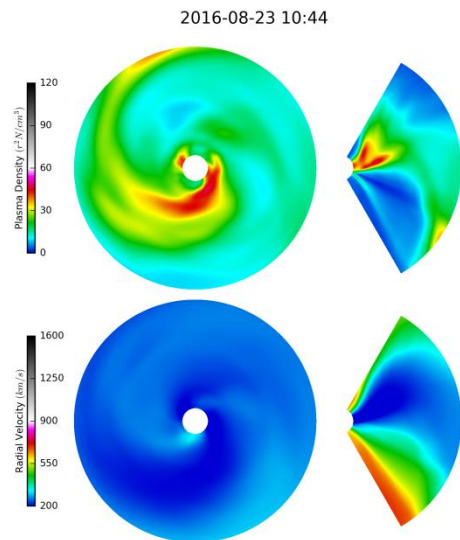
Collaboration between the **University of Michigan** and the **University of Warwick (UK)**

Start from GONG magnetogram data at the Sun

Use SWMF code AWSoM¹

Predict conditions at ~25 Solar radii

Couple to spherical MHD code SWIFT
Propagate wind conditions out to L1 and Earth



The simulated solar wind properties along the Earth orbit and the OMNI data during CR2123

1. B van der Holst et al. ApJ 782, 81 (2014)

Forecast of geomagnetic indices

Geomagnetic activity expressed in terms of geomagnetic indices such as Dst, Kp, or AE

Indices are used as inputs to numerical models for radiation environment

Methodologies used

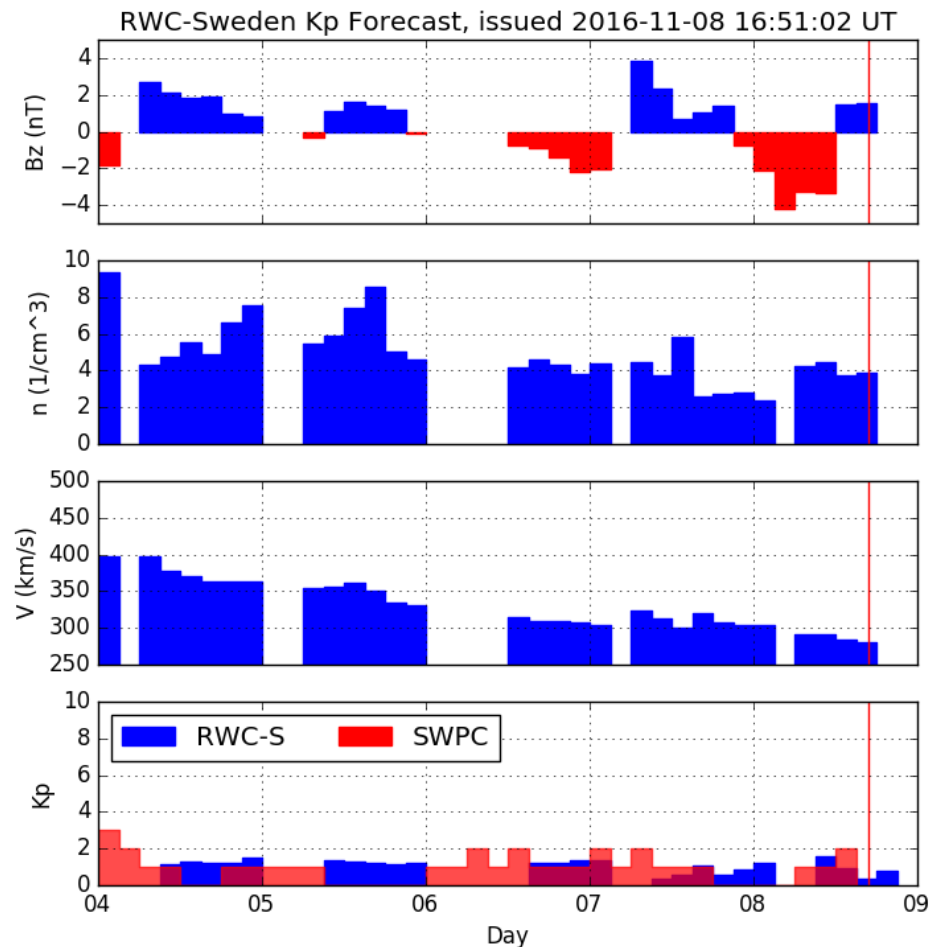
Neural nets (IRF)

NARMAX (U. Sheffield)

NARMAX + Lyapunov exponents

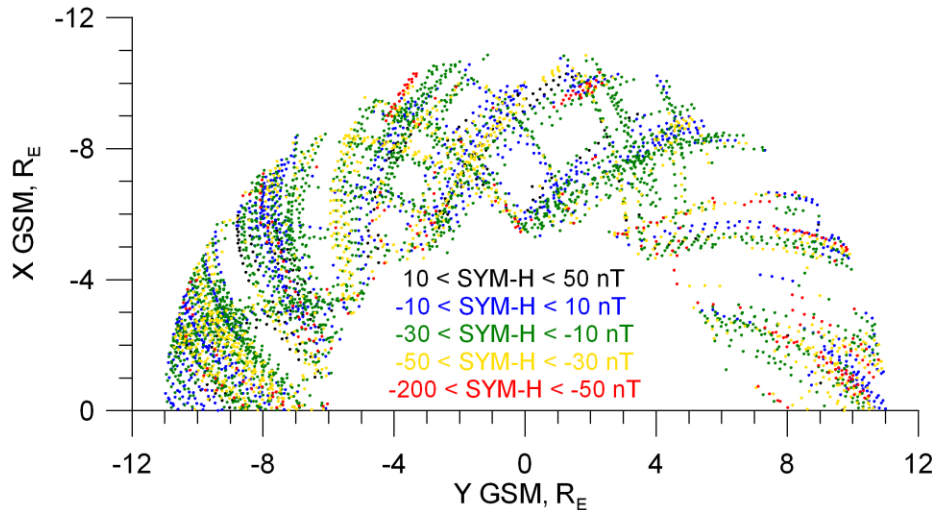
Current status – a review of current online models performed as well as study of methods to assess quality

**Swedish Institute of Space Physics,
Lund, Sweden**
(<http://www.lund.irf.se/rwc/>)



New empirical plasma sheet model

Dubyagin et al., JGR, 2016



Analysed THEMIS data 6–11 Re
 Data: THEMIS A, D, E probes;
 ESA electrons: 30eV - 30 keV;
 SST electrons ~25 keV - 300 keV

Density model: 2 input parameters

- (1) Solar wind proton density
- (2) IMF southward component

Temperature model: 3 input parameters

- (1) Solar wind velocity
- (2) IMF southward component
- (3) IMF northward component

Electron density model: 7 coefficients

$$N_e = 1.23 - 1.01 \cdot r + 0.874 \cdot r \phi^2 - 0.82 \cdot \phi^2$$

positive → +0.392 · N_{SW}
 positive → + (0.521 - 0.474 · r) · B_S

Electron temperature model: 9 coefficients

$$T_e = [-0.0215 - 0.426 \cdot \phi$$

positive → +0.874 · V_{SW}
 positive → + (0.587 - 0.538 · r φ²) · B_S^{0.32}
 negative → -0.489 · r · B_N^{0.36}]^{2.31}

Both models show very good performance

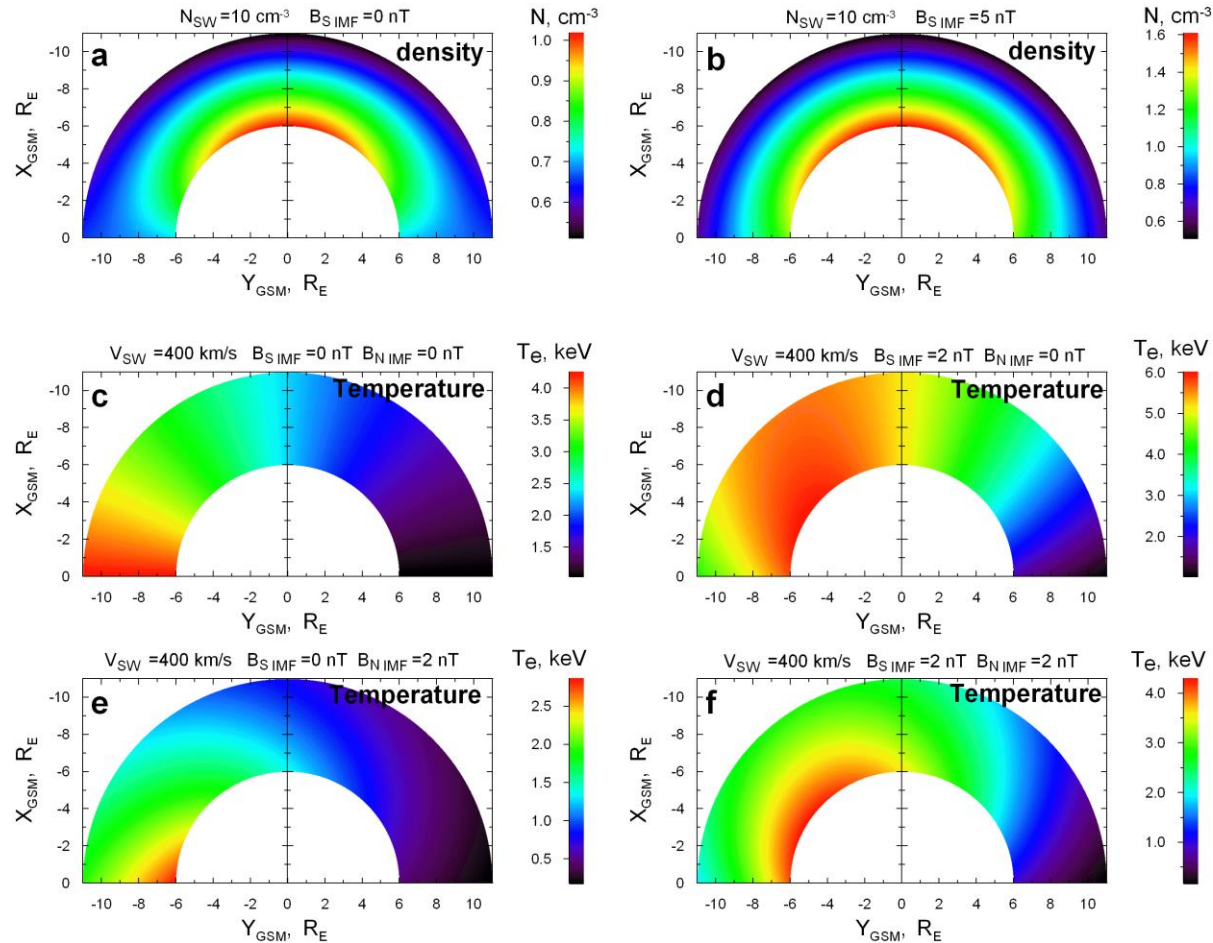
Density: C.C.=0.82; RMS = 0.23 cm⁻³

Temperature: C.C.=0.75; RMS = 2.6 keV



New empirical plasma sheet model

Dubyagin et al., JGR, 2016



Note asymmetry in electron temperature

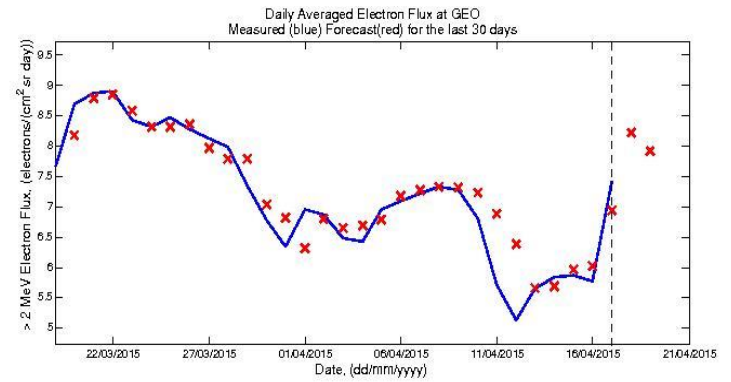
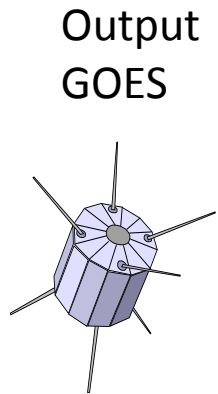
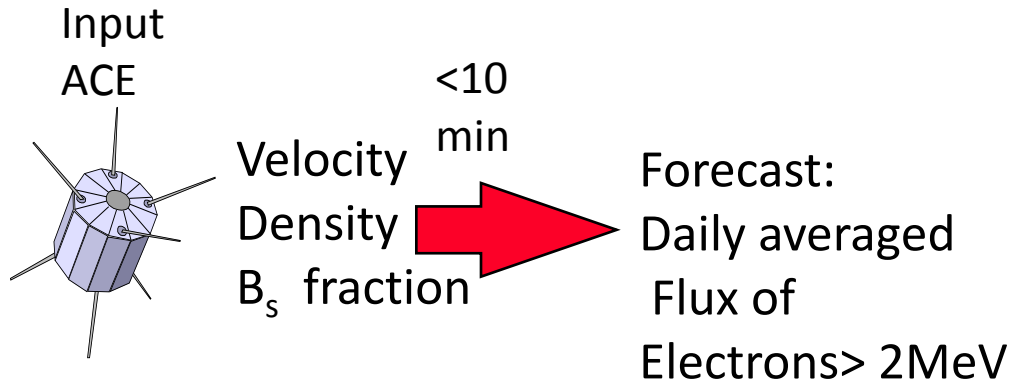
Model output can be used as a source for modelling inward transport of $< 150 \text{ keV}$ electrons



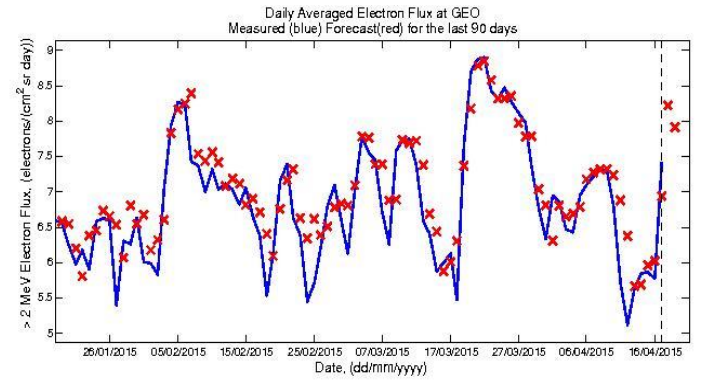
Online Forecasts – SNB³GEO

The one day ahead forecasts of the relativistic electron fluxes with energies greater than 2 MeV at GEO has been developed in Sheffield and is available in real time:

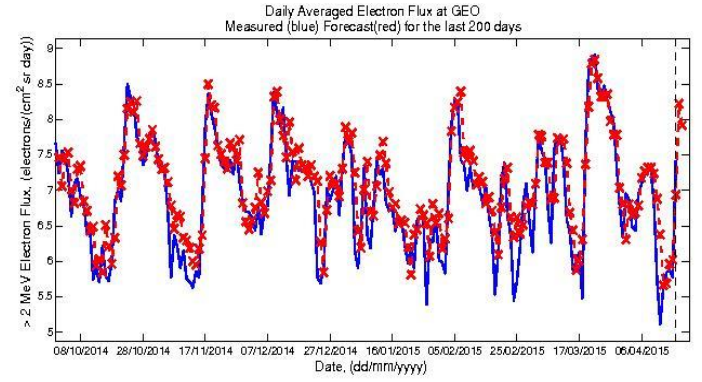
<http://ssg.group.shef.ac.uk/ssg2013/UOSSW/2MeV/EF.html>



Past 90 days



Past 200 days



Comparison of REFM and SNB³GEO forecasts

Balikhin et al., Space Weather, 2016

Model	Prediction Efficiency Flux	Correlation Flux	Prediction Efficiency Log Flux	Correlation Log Flux
REFM	-1.31	0.73	0.70	0.85
SNB ³ GEO	0.63	0.82	0.77	0.89



Experience from first EU-funded project: What end-users want is Traffic light

The screenshot shows the homepage of the SPACECAST website. At the top, there is a navigation bar with the SPACECAST logo and the European Union flag. Below this is a dark blue navigation menu with links for High-Energy Electron Forecasts, Low-Energy Electron Nowcasts, Proton Radiation Dose, Ground Based Observations, Archive, and Solar Energetic Particles. The main content area features a welcome message and a table titled "SPACECAST Satellite Risk (updated hourly)". The table shows risk levels for different orbit types and charging types. Below the table, the solar proton dose rate is indicated as "Low". On the left side, there is a sidebar with a "Home" menu and a login form.

Home

- SPACECAST Project
- News
- Publications
- Links
- Background
- How we ...
- Models
- Background
- Acknowledgements
- Contact us
- Login

Username:

Password:

[Register for an account](#)

High-Energy Electron Forecasts **Low-Energy Electron Nowcasts** **Proton Radiation Dose** **Ground Based Observations** **Archive** **Solar Energetic Particles**

Welcome to the SPACECAST web site, a resource providing support for satellite operators, designers and insurers, and information for the general public. SPACECAST is a Collaborative Project funded by the European Union Framework 7 programme to help protect satellites on orbit by modelling and forecasting particle radiation.

SPACECAST Satellite Risk (updated hourly)

	Internal charging	Surface charging
Geostationary orbit	Low	Low
Galileo/GPS orbit	Medium	Low
Slot region	Low	Low

Solar proton dose rate: **Low**

Determination of the 1 in N year event

- Our major objective is to determine the 1 in N year space weather event
- The flux that is exceeded on average once every N years can be expressed in terms of the fitted parameters σ and ξ as:

$$x_N = u + (\sigma/\xi)(Nn_d n_c/n_{tot})^\xi - 1))$$

where n_d is the number of data points in a given year, n_c is the number of cluster maxima and n_{tot} is the total number of data points

- A plot of x_N against N is known as a return level plot